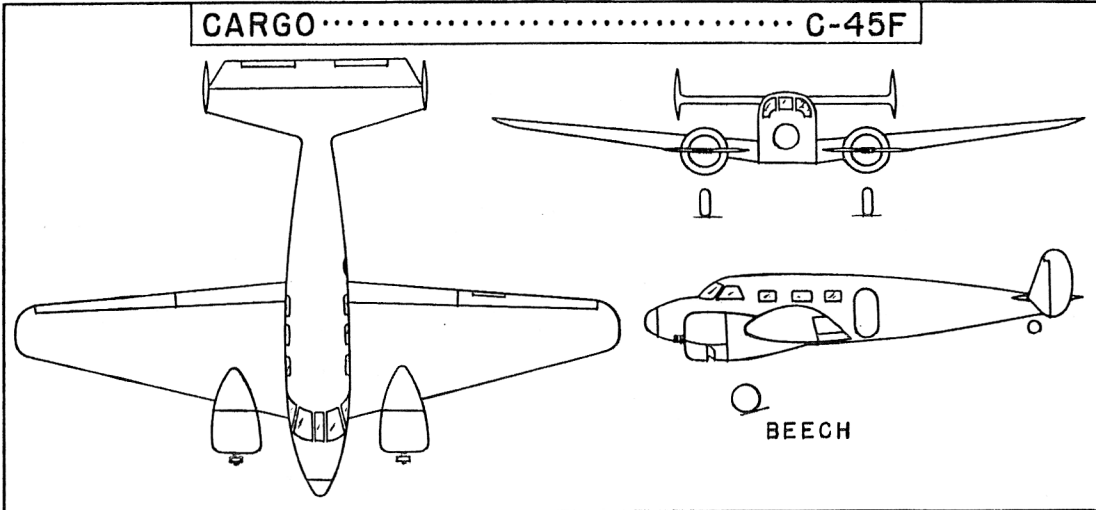


Characteristics Summary

CARGO G-45F



Wing area 349 sq ft Length 34.2 ft
 Span 47.6 ft Height 9.3 ft

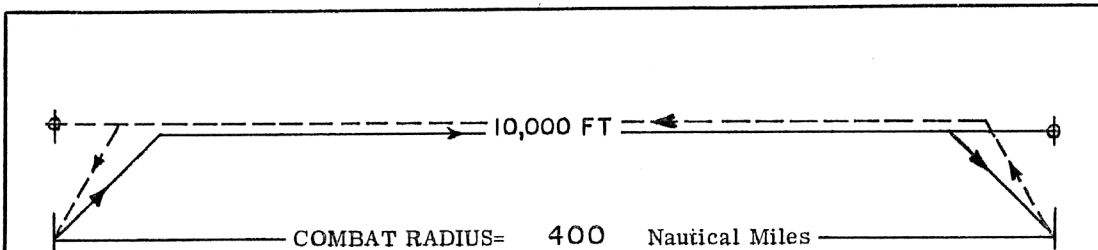
A V A I L A B I L I T Y			P R O C U R E M E N T			
Number available			Number to be delivered in fiscal years			
ACTIVE	RESERVE	TOTAL				

S T A T U S

1. First acceptance: March 1940
2. Production completed: August 1945
3. Navy designation: JRB-4

P O W E R P L A N T	F E A T U R E S	A R M A M E N T						
(2) R-985-AN-1 or AN-3 Pratt-Whitney ENGINE RATINGS <table style="width: 100%; border: none;"> <tr> <td></td> <td style="text-align: center;">BHP - RPM - ALT</td> </tr> <tr> <td>T.O:</td> <td>450 - 2300 - S.L.</td> </tr> <tr> <td>Nor:</td> <td>400 - 2200 - 5100</td> </tr> </table>		BHP - RPM - ALT	T.O:	450 - 2300 - S.L.	Nor:	400 - 2200 - 5100	Crew: 2 Passengers: 5 Prop Anti-icing Boot De-icer Life Rafts (2) (prov) Auto-Pilot Max Fuel Cap: 247 gal	NONE
	BHP - RPM - ALT							
T.O:	450 - 2300 - S.L.							
Nor:	400 - 2200 - 5100							

Characteristics Summary Basic Mission C-45F



P E R F O R M A N C E		
COMBAT RADIUS	COMBAT RANGE	COMBAT SPEED
400 naut. mi with 1000 lb payload at 111 knots avg. in 7.5 hours.	635 ⁽²⁾ naut. mi with 1000 lb payload at 113 knots avg. in 5.8 hours.	192 knots at 10,000 ft alt, max power
		MAXIMUM SPEED
		195 knots at 3000 ft alt, max power
C L I M B	C E I L I N G	T A K E - O F F
1110 fpm sea level, take-off weight normal power	21,000 ft 100 fpm, take-off weight normal power	ground run 1030 ft no assist — ft assisted
1870 fpm sea level, combat weight maximum power	20,700 ft 500 fpm, combat weight maximum power	over 50 ft height 1530 ft no assist — ft assisted
L O A D	W E I G H T S	S T A L L I N G S P E E D
Passengers: 5	Empty..... 5850 lb Combat... 7040 lb Take - off 8800 lb limited by space	59 knots flaps down, take-off weight
Fuel: *238 gal protected 0 % droppable 0 % external 0 %		TIME TO CLIMB —

N O T E S

- PERFORMANCE BASIS:
 - Flight test
 - Fuel density: 6.0 lb/gal
 - In computing Radius and Range, specific fuel consumptions have been increased 5% to allow for variation of fuel flow in service aircraft.
 - Performance shown above for max power is based on 450 BHP @ 2300 RPM @ 3000 ft.
 - Nose auxiliary fuel tank must remain full if two aft passenger seats are occupied
 - REVISION BASIS: To show engine ratings and changes in "Status", "Features" and "Notes" blocks.
- * Note: Majority of aircraft now have 47 gal nose tank. (Total fuel 247 gal). A few aircraft will retain 38.5 gal nose tank. (Total fuel 238.5 gal). Above performance is based on 238 gal fuel.